

Constrained Safe Cooperative Maneuvering of Autonomous Surface Vehicles: A Control Barrier Function Approach

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Abstract—This article investigates a constrained safe cooperative maneuvering method for a group of autonomous surface vehicles (ASVs) with performance-quantized indices in an obstacle-loaded environment. Specifically, an avoidance-tolerant prescribed performance (ATPP) with one-sided tunnel bounds is designed to predetermine the cooperative maneuvering performance of multiple ASVs. Next, an auxiliary system is constructed to modify performance bounds of ATPP for tolerating possible collision avoidance actions of ASVs. In the guidance loop, nominal surge and yaw guidance laws are developed using the ATPP-based transformed relative distance and heading errors. A barrier-certified yaw velocity protocol is proposed by formulating a quadratic optimization problem, which unifies the nominal yaw guidance law and CBF-based collision-free constraints. In the control loop, two prescribed-time disturbance observers (PTDOs) are devised to estimate unknown external disturbances in the surge and yaw directions. The antidisturbance control laws are designed to track the guidance signals. By the stability and safety analysis, it is proved that error signals of the proposed closed-loop system are bounded and the multi-ASV system is input-to-state safe. Finally, simulation results are used to demonstrate the effectiveness of the presented constrained safe cooperative maneuvering method.

Index Terms—Avoidance-tolerant prescribed performance (ATPP), control barrier function (CBF), cooperative maneuvering, safety-critical control, underactuated autonomous surface vehicles.

Manuscript received 30 April 2023; revised 15 October 2023; accepted 15 December 2023. This work was supported in part by the National Key Research and Development Program of China under Grant 2022ZD0119901; in part by the Shanghai Science and Technology Program under Grant 22015810300; in part by the Hainan Province Science and Technology Special Fund under Grant ZDYF2021GXJS041; in part by the National Natural Science Foundation of China under Grant U2141234, Grant 62073217, and Grant 52201369; in part by the China Post-Doctoral Science Foundation under Grant 2022M722053; and in part by the Oceanic Interdisciplinary Program of Shanghai Jiao Tong University under Grant SL2022PT112. This article was recommended by Associate Editor J. Wang. (Corresponding authors: Yibo Zhang; Weidong Zhang.)

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Color versions of one or more figures in this article are available at <https://doi.org/10.1109/TSMC.2023.3345847>.

Digital Object Identifier 10.1109/TSMC.2023.3345847

I. INTRODUCTION

IN RECENT years, autonomous surface vehicle (ASV), as a kind of intelligent marine vehicles without human intervention, has gained increasing attentions due to promising maritime applications, such as ocean exploration, surveillance, mapping, and transportation, to name a few [1], [2], [3], [4], [5]. To enhance the capability and effectiveness of accomplishing missions, a group of ASVs are driven to implement the cooperative operation for a collaborative behavior by sharing their own information with other individuals. The typical cooperative patterns include consensus [6], containment [7], flocking [8], target enclosing [9], target tracking [10], formation [11], [12], etc. The primary challenge of cooperative control lies in devising the control protocol for every ASV to establish and maintain a predefined geometric configuration.

For ASVs in the sea, it is difficult to achieve the cooperation task owing to model nonlinearities, parameter uncertainties, and external disturbances. Consequently, the existing research results on cooperative control have been investigated in [13], [14], [15], [16], [17], and [18]. In [13] and [14], the spatial-temporal cooperative control schemes with path maneuvering techniques are proposed based on the approximated vehicle kinetics from echo state network (ESN)-based estimators. In [15], the approximation speed of ESNs is improved by integrating the accelerated learning technique. Besides the adaptive approximation methods [13], [14], [15], extended state observers (ESOs) are also employed to not only estimate disturbances but also recovery states. In [16] and [17], finite-time ESOs (FTESOs) are presented to obtain the fast estimation ability. In [16], an FTESO-based even-triggered control scheme is derived to degrade the communication burden of multi-ASV containment task. In [17], the robust exact differentiator-based FTESO is developed to estimate the velocity information of leader vessel within finite time. In [18], the cooperative dynamic positioning of ASVs with multiple operating points is performed by developing a network-based control scheme under the fixed topology. The aforementioned controllers can force cooperative errors to converge to a residual set. Note that convergence rate and state-steady accuracy of cooperative errors are not explicitly preconfigured to a safety region, which may degrade the multi-ASV system reliability.

To explicitly depict the performance constraints, [19] proposes the prescribed performance control (PPC) for MIMO nonlinear systems to obtain the desired transient and

steady-state indices by predetermining behavioral functions. There are many research results on PPC applications to marine vehicle systems [20], [21], [22], [23], [24], [25], [26], [27], [28]. In [20], by using disturbance observers (DOs), a PPC-based distributed control method is proposed to perform the containment formation of multiple ASVs under unknown external disturbances. In [21] and [22], high-gain observers-based adaptive output-feedback control schemes are presented for ASVs with unavailable velocities to achieve performance-prescribed trajectory tracking. In [23] and [24], input and output constraints of ASVs are considered in the distributed cooperative control by constructing the adaptive fuzzy state observers. In [25], [26], and [27], the trajectory tracking control of ASVs with prescribed performance and actuator faults are derived using the fault compensation approaches. It should be pointed out that these methods do not address the safety problem of multi-ASV system, i.e., vehicle avoidance and obstacle avoidance.

In an obstacle-loaded ocean, the safe cooperative operation of multiple ASVs is critical and challenging due to encountering vehicles and dynamic/stationary obstacles. Some collision avoidance strategies have been reported, including differentiable function [29], vector field [30], potential function [31], [32], [33], PPC [34], [35], and control barrier functions (CBFs) [36], [37], [38], [39], to name a few. In [30], a topology-switched containment maneuvering method is devised using the vector field approach for tackling inter-ASV collisions. In [36], [37], and [39], CBF-based collision-free protocols are derived by formulating an optimization problem to avoid encountering ASVs and obstacles. In [34] and [35], collision-free constraints are incorporated into the PPC control framework, and a leader-follower formation method is designed to not only obtain guaranteed performance but also avoid intervehicle collisions. Along with precision and intervehicle collision avoidance, [32], [33] also avoid collision with stationary obstacles by introducing potential functions. It is worth noting that these PPC methods [32], [33], [34], [35] cannot ensure the safety of multi-ASV formation subject to encountering ASVs and dynamic obstacles. Although CBF-based methods can deal with aforementioned collision-free scenarios, transient, and state-steady performance of tracking errors are ignored. For existing PPC and CBF methods, it is a difficult task to simultaneously meet performance and safety constraints because inflexible bounds have poor tolerance for fluctuating errors.

Motivated by the above discussions, this article aims to develop a constrained safe cooperative maneuvering method for multiple underactuated ASVs subject to dynamic and stationary obstacles. The main contributions of this article are detailed as follows.

- 1) In contrast to cooperative control methods [13], [14], [15], [16], [17], [18], [20], [21], [22], [23], [24], [25], [26], [27], [28], [29], [30], [31], [36], [37], [38], [39] without simultaneously considering output and safety constraints, this article develops a constrained safe cooperative maneuvering method for multiple ASVs in an obstacle-loaded environment. Different from [6], [32], [33], [34], [35], the cooperation safety of the multi-ASV

system subject to dynamic obstacles is also guaranteed using the developed yaw velocity protocol based on second-order CBFs.

- 2) In contrast to PPC and TPP methods [20], [21], [22], [23], [24], [25], [26], [27], [28], [32], [33], [34], [35], where behavioral functions are possibly violated due to collision avoidance, this article designs an avoidance-tolerant prescribed performance (ATPP) capable of enlarging or recovering error bounds. The designed ATPP establishes a tradeoff mechanism between tracking errors and output constraints by constructing auxiliary systems associated with safety constraints.
- 3) In contrast to DOs and ESOs proposed in [29], [33], [40], [41], [42], [43], [44], [45], [46] with asymptotic, finite-time, or fixed-time convergence abilities, the prescribed-time DOs (PTDOs) are designed to estimate unknown disturbances within a predefined constant. The setting time of designed PTDOs is independent on initial values.

This article is organized as follows. Section II introduces preliminaries and problem formulation. Section III designs the constrained safe cooperative maneuvering controller. Section IV gives the stability analysis. Section V provides simulation results. Section VI concludes this article.

II. PRELIMINARIES AND PROBLEM FORMULATION

A. Preliminaries

1) *Notations*: Throughout this article, \mathfrak{R}^+ , \mathfrak{R}^m , and $\mathfrak{R}^{m \times n}$ represents a positive real space, an m -dimensional Euclidean space, and an $m \times n$ -dimensional Euclidean space, respectively. 0_m and $0_{m \times n}$ denote an m -dimensional zero vector and a $m \times n$ -dimensional zero matrix, respectively. $\text{diag}\{\cdot\}$ is a block-diagonal matrix. $\|\cdot\|$ denotes the Euclidean norm of a vector. $\bar{\lambda}(\cdot)$ and $\underline{\lambda}(\cdot)$ represent the minimum and maximum of a symmetric matrix. \otimes denotes a Kronecker product.

2) *Graph Theory*: This article considers a system containing M ASVs, $(N - M)$ virtual leaders, and one super leader. The information transmission among them can be depicted via a graph $\mathcal{G} = \{\mathcal{N}, \mathcal{V}, \mathcal{E}, \mathcal{A}\}$. $\mathcal{N} = \{n_0, \dots, n_N\}$ denotes the set of all nodes. $\mathcal{V} = \{\mathcal{V}^F, \mathcal{V}^L, \mathcal{V}^S\}$ is a vertex set with $\mathcal{V}^F = \{n_1, \dots, n_M\}$, $\mathcal{V}^L = \{n_{M+1}, \dots, n_N\}$, and $\mathcal{V}^S = \{n_0\}$. $\mathcal{E} = \{(n_i, n_j) \in \mathcal{V} \times \mathcal{V}\}$ is an edge set to describe the information flow among nodes n_i and n_j . To depict the neighboring relationship, the neighborhood set of node n_i is defined as $\mathcal{N}_i = \{\mathcal{N}_i^F, \mathcal{N}_i^L, \mathcal{N}_i^S\}$ with $\mathcal{N}_i^F = \{n_j \in \mathcal{V}^F | (n_i, n_j) \in \mathcal{E}\}$, $\mathcal{N}_i^L = \{n_j \in \mathcal{V}^L | (n_i, n_j) \in \mathcal{E}\}$, and $\mathcal{N}_i^S = \{n_j \in \mathcal{V}^S | (n_i, n_j) \in \mathcal{E}\}$. $\mathcal{A} \in \mathfrak{R}^{(N+1) \times (N+1)}$ is an adjacency matrix with $\mathcal{A} = [a_{ij}]$. If node n_i can acquire the information flow from node n_j , i.e., $(n_i, n_j) \in \mathcal{E}$, $a_{ij} = 1$, otherwise, $a_{ij} = 0$. The graph \mathcal{V} is called as the undirected graph if $a_{ij} = a_{ji}$. $\mathcal{D} \in \mathfrak{R}^{(N+1) \times (N+1)}$ is a degree matrix defined by $\mathcal{D} = \text{diag}\{d_i\}$ with $d_i = \sum_{j \in \mathcal{N}_i} a_{ij}$. A Laplacian matrix $\mathcal{L} \in \mathfrak{R}^{(N+1) \times (N+1)}$ is devised as follows:

$$\mathcal{L} = \mathcal{D} - \mathcal{A} = \begin{bmatrix} 0 & 0_M^T & \mathcal{L}_3 \\ 0_M & \mathcal{L}_1 & \mathcal{L}_2 \\ \mathcal{L}_3^T & \mathcal{L}_2^T & \mathcal{L}_0 \end{bmatrix} \quad (1)$$

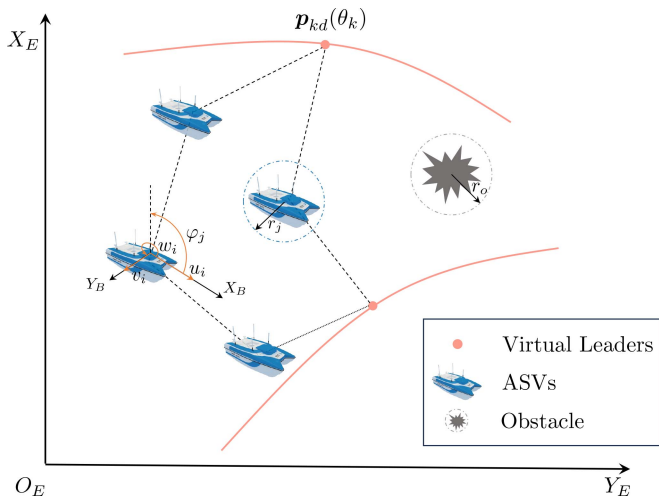


Fig. 1. Illustration for safe cooperative maneuvering of a group of ASVs in an obstacle-loaded environment.

where $\mathcal{L}_0 \in \mathbb{R}^{(N-M) \times (N-M)}$, $\mathcal{L}_1 \in \mathbb{R}^{M \times M}$, $\mathcal{L}_2 \in \mathbb{R}^{M \times (N-M)}$, and $\mathcal{L}_3 \in \mathbb{R}^{1 \times (N-M)}$.

3) *Prescribed-Time Control*: To obtain the prescribed-time convergence property, a time-varying scaling function is defined as follows [47]:

$$\mu(t) = \begin{cases} \frac{T^b}{(T+t_0-t)^b}, & t_0 \leq t < t_1 \\ 1, & t_1 \leq t \end{cases} \quad (2)$$

where $b \geq 2$ and $t_1 = t_0 + T$ are any user-set constants. Note that $\mu(t)^{-\alpha}$ ($\alpha \in \mathbb{R}^+$) is monotonically decreasing for $t \in [t_0, t_1)$, $\mu(t_0)^{-\alpha} = 1$ and $\lim_{t \rightarrow t_1^-} \mu(t)^{-\alpha} = 0$. Besides, one has $\dot{\mu}(t) = b/T\mu^{1+1/b}$ for $t \in [t_0, t_1)$ and $\dot{\mu}(t) = 0$ for $t \in [t_1, \infty)$.

Lemma 1 [47]: Consider a system $\dot{x}(t) = f(x(t), t)$, $t \in \mathbb{R}^+$, $x_0 = x(t_0)$. Construct a continuously differentiable function $V(x(t), t) : U \times \mathbb{R}^+ \mapsto \mathbb{R}$ with $U \subset \mathbb{R}^m$ being a domain including the origin. If there exists a constant $k \in \mathbb{R}^+$ such that $V(0, t) = 0$, $V(x(t), t) > 0$ in $U - \{0\}$, and $\dot{V} = -kV - 2\dot{\mu}/\mu V$ in U on $[t_0, \infty)$, then the origin of system $\dot{x}(t) = f(x(t), t)$ is prescribed-time stable. It holds $V(t) \leq \mu^{-2}e^{-k(t-t_0)}V(t_0)$ for $t \in [t_0, t_1)$, and $V(t) \equiv 0$ for $t \in [t_1, \infty)$.

B. Problem Formulation

As shown in Fig. 1, two reference frames, Earth-fixed and body-fixed reference frames, are employed to describe the horizontal motion of M ASVs. For underactuated ASVs, it usually ignores the heave, pitch, and roll due to the X_B - Z_B -plane symmetry [29]. Then, the model dynamics of i th ASV is expressed as follows:

$$\begin{cases} \dot{p}_i = \mathcal{R}(\varphi_i)v_i \\ \dot{\varphi}_i = w_i \\ \dot{u}_i = \frac{m_{iv}}{m_{iu}}v_iw_i - \frac{d_{iu}}{m_{iu}}u_i + \frac{1}{m_{iu}}(\tau_{iu} + \tau_{iu}^d) \\ \dot{v}_i = -\frac{m_{iu}}{m_{iv}}u_iw_i - \frac{d_{iv}}{m_{iv}}v_i + \frac{1}{m_{iv}}\tau_{iv}^d \\ \dot{w}_i = \frac{m_{iu}-m_{iv}}{m_{iw}}u_iv_i - \frac{d_{iw}}{m_{iw}}w_i + \frac{1}{m_{iw}}(\tau_{iw} + \tau_{iw}^d) \end{cases} \quad (3)$$

where $i = 1, \dots, M$; $p_i = [x_i, y_i]^T$ and φ_i denote the position and yaw angle; $v_i = [u_i, v_i]^T$ and w_i represent the velocity

information in the surge, sway, and yaw axes; m_{iu} , m_{iv} , and m_{iw} are the masses and inertia, respectively; d_{iu} , d_{iv} , and d_{iw} are the damping parameters; τ_{iu} and τ_{iv} stand for the surge force and yaw moment, respectively; τ_{iu}^d , τ_{iv}^d , and τ_{iw}^d denote the external disturbances; $\mathcal{R}(\varphi_i)$ is a rotation matrix with the following property.

Property 1 [28]: The rotation matrix \mathcal{R}_i is orthogonal satisfying $\|\mathcal{R}_i\| = 1$, $\mathcal{R}_i^{-1} = \mathcal{R}_i^T$, and $\dot{\mathcal{R}}_i = w_i S \mathcal{R}_i$ with

$$\mathcal{R}_i = \begin{bmatrix} \cos\varphi_i & -\sin\varphi_i \\ \sin\varphi_i & \cos\varphi_i \end{bmatrix} \text{ and } S = \begin{bmatrix} 0 & -1 \\ 1 & 0 \end{bmatrix}.$$

Assume that there exist $N_o \geq 0$ obstacles modeled as a Euclidean plane zone $\mathcal{O}_o \in \mathbb{R}^2$ with the center $p_o \in \mathbb{R}^2$, $o = 1, \dots, N_o$. The boundary of zone \mathcal{O}_o is denoted by $\partial\mathcal{O}_o \in \mathbb{R}^2$. In this article, the obstacle is assumed as to be circular with $\partial\mathcal{O}_o = \{p \in \mathbb{R}^2 \mid \|p - p_o\|^2 - r_o^2 = 0\}$, where $r_o \in \mathbb{R}^+$ represents the radius of the o th circular obstacle.

Definition 1 (Obstacle Avoidance Zone): For an open set $\mathcal{C}_{io}(p_i) = \{p_i \in \mathbb{R}^2 \mid \|p_i - p_o\|^2 < (r_o + R_{\text{obs}})^2\}$ with $R_{\text{obs}} > 0$ being a user-specified constant, the obstacle avoidance zone $\mathcal{C}_i(p_i)$ is defined by $\mathcal{C}_i(p_i) = \bigcup_{o=1}^{N_o} \mathcal{C}_{io}(p_i)$.

Aside from avoiding obstacles, each ASV must also ensure a safe distance from other ASVs to avoid collisions. Similar to obstacles, each ASV can be enclosed by a circular zone with a center p_i and a radius r_i , $i = 1, \dots, M$. Then, the ASV avoidance zone for the i th ASV is defined as follows.

Definition 2 (ASV Avoidance Zone): For an open set $\mathcal{S}_{ij}(p_i) = \{p_i \in \mathbb{R}^2 \mid \|p_i - p_j\|^2 < (r_i + r_j + R_{\text{veh}})^2\}$, $j = 1, \dots, M$, $j \neq i$ with $R_{\text{veh}} > 0$ being a user-specified constant, the ASV avoidance zone $\mathcal{S}_i(p_i)$ is defined by $\mathcal{S}_i(p_i) = \bigcup_{j=1, j \neq i}^M \mathcal{S}_{ij}(p_i)$.

Remark 1: Even if an obstacle or an ASV is not circular, possibly even having nonsmooth edges, a circle with the smallest radius enables the obstacle or ASV to be enclosed. Although the larger R_{obs} and R_{veh} are to be desired for safety purposes, the overlarge values may lead to some unnecessary sacrifices in cooperative performance. Then, the selection of these parameters is recommended to hold a tradeoff between safety and stability. The collision-free protocol designs based on circular obstacles have been widely used in multirobot systems [34], [48], [49].

In order to achieve cooperative behavior of M ASVs guided by $(N-M)$ virtual leaders, a series of parameterized paths are first defined as follows:

$$p_{kd}(\theta_k) = [x_{kd}(\theta_k), y_{kd}(\theta_k)]^T \in \mathbb{R}^2 \quad (4)$$

where $k = M+1, \dots, N$; $\theta_k \in [0, 1)$ is a path variable. Suppose that $p_{kr}(\theta_k)$ and its partial derivative $p_{kr}^{\theta_k} = \partial p_{kr}(\theta_k) / \partial \theta_k$ are bounded. Some necessary assumptions are given as follows.

Assumption 1 [14]: The graph \mathcal{G} is undirected and has a spanning tree with the super leader being a root node.

Assumption 2 [43]: The external disturbances for each ASV are bounded, i.e., $\|\tau_i^d\| \leq \bar{\tau}_i^d \in \mathbb{R}^+$ with $\tau_i^d = [\tau_{iu}^d, \tau_{iv}^d, \tau_{iw}^d]^T$.

For practical ASV systems, external disturbances are caused by winds, waves, and currents with the limited energy. Thus, Assumption 2 is reasonable.

In an obstacle-loaded environment, all ASVs not only execute the graph-defined cooperative pattern but also ensure their own safety. In what follows, this article aims to develop a performance-prescribed collision-free cooperative maneuvering method for underactuated ASVs such that the following objectives can be achieved.

- 1) *Geometric Objective*: Conduct each ASV to converge to a convex hull spanned by virtual leaders, i.e.,

$$\left\| p_i - \sum_{k=M+1}^N \beta_{ki} p_{kd}(\theta_k) \right\| \leq \varepsilon_{ip} \in \mathfrak{R}^+ \quad (5)$$

where $\beta_{ki} \in \mathfrak{R}^+$ is a constant with $\sum_{k=M+1}^N \beta_{ki} = 1$.

- 2) *Dynamic Objective*: Force each virtual leader to complete the velocity task and phase task, i.e.,

$$\begin{aligned} \text{Velocity task: } |\dot{\theta}_k - v_s| &\leq \varepsilon_{k\theta 1} \in \mathfrak{R}^+ \\ \text{Phase task: } |\theta_k - \theta_0 - \mathcal{P}_k| &\leq \varepsilon_{k\theta 2} \in \mathfrak{R}^+ \end{aligned} \quad (6)$$

where $k = M + 1, \dots, N$; $v_s \in \mathfrak{R}$ denotes the update velocity of path variable θ_0 for super leader with $\dot{\theta}_0 = v_s$; $\mathcal{P}_k \in \mathfrak{R}$ is a user-predefined deviation.

- 3) *Avoidance Collision*: Hold each ASV away from ASV avoidance zone, i.e.,

$$p_i \notin \mathcal{S}_{ij}(p_j), \quad j = 1, \dots, M, \quad j \neq i \quad \forall t \geq t_0. \quad (7)$$

- 4) *Avoidance Obstacles*: Keep each ASV away from the obstacle avoidance zone, i.e.,

$$p_i \notin \mathcal{C}_{io}(p_i), \quad o = 1, \dots, N_o \quad \forall t \geq t_0. \quad (8)$$

III. CONSTRAINED SAFE CONTROLLER DESIGN

In this section, we present a constrained safe cooperative maneuvering method for underactuated ASVs by developing performance-prescribed guidance laws, a barrier-certified yaw velocity protocol, and PTDO-based control laws.

A. Performance-Prescribed Guidance Laws

To achieve the cooperative maneuvering guided by multiple virtual leaders, a distributed error $p_{ie} = [x_{ie}, y_{ie}]^T$ for the i th ASV is defined as follows:

$$p_{ie} = \sum_{j \in \mathcal{N}_i^F} a_{ij}(p_i - p_j) + \sum_{k \in \mathcal{N}_i^L} a_{ik}(p_i - p_{kd}(\theta_k)). \quad (9)$$

With the error vector p_{ie} , a guidance heading angle $\varphi_{id} \in (-\pi, \pi]$ is obtained by

$$\varphi_{id} = \text{atan2}(y_{ie}, x_{ie}) - \pi \text{sign}(y_{ie}) \quad (10)$$

where $\text{atan2}(\cdot, \cdot) \in (-\pi, \pi]$ is an inverse tangent function.

Then, the relative distance z_{i1} and the heading error z_{i2} are presented as follows:

$$\begin{cases} z_{i1} = \|p_{ie}\| \\ z_{i2} = \varphi_i - \varphi_{id}. \end{cases} \quad (11)$$

In order to allow users to preset performance indices of cooperative maneuvering, we propose the ATPP to force errors z_{i1} and z_{i2} with respect to

$$-z_{ib}^l - \lambda_{ib}^l \leq z_{ib} \leq z_{ib}^r + \lambda_{ib}^r, \quad b = 1, 2 \quad (12)$$

with

$$\begin{aligned} z_{ib}^r &= [\delta_{ib}^r + \text{sign}(z_{ib}(t_0))] \rho_{ib} - \rho_{i\infty} \text{sign}(z_{ib}(t_0)) \\ z_{ib}^l &= [\delta_{ib}^l - \text{sign}(z_{ib}(t_0))] \rho_{ib} + \rho_{i\infty} \text{sign}(z_{ib}(t_0)) \end{aligned} \quad (13)$$

where $0 \leq \delta_{ib}^r, \delta_{ib}^l \leq 1$ are scale parameters; $\rho_{ib}(t)$ is a monotonically decreasing function defined as $\rho_{ib} = (\rho_{ib,0} - \rho_{ib,\infty})e^{-\iota_{ib}(t-t_0)} + \rho_{ib,\infty}$ with $\rho_{ib,0} = \rho_{ib}(t_0)$, $\rho_{ib,\infty} = \lim_{t \rightarrow \infty} \rho_{ib}(t)$, $\rho_{ib,0} > \rho_{ib,\infty} > 0$, and $\iota_{ib} > 0$; $\lambda_{ib}^r \geq 0$ and $\lambda_{ib}^l \geq 0$ are non-negative adjusted variables to eliminate the limitation of inflexible bounds. Note that the initial errors $z_{i1}(t_0)$ and $z_{i2}(t_0)$ satisfy the ATPP constraints, i.e., $0 < z_{i1}(t_0) < z_{i1}^r(t_0)$ and $-z_{i2}^r(t_0) < z_{i2}(t_0) < z_{i2}^l(t_0)$ for $p_i(t_0) \notin \mathcal{C}_i(p_i) \cup \mathcal{S}_i(p_i)$ with $i \in \mathcal{V}^F$.

Based on the PPC methodology [19], a nonlinear mapping from original error z_{ib} to transformed variable ξ_{ib} is formulated as follows:

$$z_{ib} = 0.5(\bar{z}_{ib}^r + \underline{z}_{ib}^l) \Upsilon(\xi_{ib}) + 0.5(\bar{z}_{ib}^r - \underline{z}_{ib}^l) \quad (14)$$

where $\bar{z}_{ib}^r = z_{ib}^r + \lambda_{ib}^r$ and $\underline{z}_{ib}^l = z_{ib}^l + \lambda_{ib}^l$; $\Upsilon(\xi_{ib}) : (-\infty, \infty) \mapsto [-1, 1]$ is a smooth and monotonically increasing function.

With the error transformation function $\Upsilon(\xi_{ib}) = 2/\pi \arctan(\xi_{ib})$, the corresponding unconstrained variables ξ_{ib} for z_{ib} is yielded by (14)

$$\xi_{ib} = \Upsilon^{-1}\left(z_{ib}, \bar{z}_{ib}^r, \underline{z}_{ib}^l\right) = \tan\left(\frac{\pi}{2} \frac{2z_{ib} - \bar{z}_{ib}^r + \underline{z}_{ib}^l}{\bar{z}_{ib}^r + \underline{z}_{ib}^l}\right). \quad (15)$$

Let $\mathcal{F}_{ib} = \partial \xi_{ib} / \partial z_{ib}$, $\mathcal{F}_{ib}^r = \partial \xi_{ib} / \partial \bar{z}_{ib}^r$, and $\mathcal{F}_{ib}^l = \partial \xi_{ib} / \partial \underline{z}_{ib}^l$, and take the derivative of ξ_{ib} along (15) as follows:

$$\dot{\xi}_{ib} = \mathcal{F}_{ib} \dot{z}_{ib} + \mathcal{F}_{ib}^r (\dot{\bar{z}}_{ib}^r + \dot{\lambda}_{ib}^r) + \mathcal{F}_{ib}^l (\dot{\underline{z}}_{ib}^l + \dot{\lambda}_{ib}^l). \quad (16)$$

Since $p_{ie}/\|p_{ie}\| = [\cos(\varphi_{id} + \pi \text{sign}(y_{ie})), \sin(\varphi_{id} + \pi \text{sign}(y_{ie}))]^T = [-\cos(\varphi_{id}), -\sin(\varphi_{id})]^T$, and $\cos(z_{i2}) = 1 - 2 \sin^2(z_{i2}/2)$, the derivatives of z_{i1} and z_{i2} are taken as follows:

$$\begin{cases} \dot{z}_{i1} = -d_i \left[u_i - 2u_i \sin^2\left(\frac{z_{i2}}{2}\right) - v_i \sin(z_{i2}) \right] \\ \quad - \frac{p_{ie}^T}{\|p_{ie}\|} \left[\sum_{j \in \mathcal{N}_i^F} a_{ij} \mathcal{R}_j v_j + \sum_{k \in \mathcal{N}_i^L} a_{ik} p_{kd}^{\theta_k} v_s \right. \\ \quad \left. - \sum_{k \in \mathcal{N}_i^L} a_{ik} p_{kd}^{\theta_k} \omega_k \right] \\ \dot{z}_{i2} = w_i - \dot{\varphi}_{id} \end{cases} \quad (17)$$

where $d_i = \sum_{j \in \mathcal{N}_i^F} a_{ij} + \sum_{k \in \mathcal{N}_i^L} a_{ik}$.

In what follows, we design a first-order auxiliary system to update variables λ_{ib}^r and λ_{ib}^l . Considering the unknown direction of error changes caused by collision avoidance, the auxiliary system is constructed as follows:

$$\begin{cases} \dot{\lambda}_{ib}^r = -\mathcal{F}_{ib} / \mathcal{F}_{ib}^r (-\kappa_{ib} \lambda_{ib}^r + G_{ib}^r), \quad \lambda_{ib}^r(t_0) = 0 \\ \dot{\lambda}_{ib}^l = \mathcal{F}_{ib} / \mathcal{F}_{ib}^l (-\kappa_{ib} \lambda_{ib}^l + G_{ib}^l), \quad \lambda_{ib}^l(t_0) = 0 \end{cases} \quad (18)$$

where κ is a positive constant; $G_{ib}^r(z_{ib}) = G_{ib}^l(z_{ib}) = \bigvee_{j=1, \dots, M, j \neq i, o=1, \dots, N_o} [\|p_i - p_o\| - (r_o + R_{\text{obs}}) < 0 \bigvee \|p_i - p_j\| - (r_i + r_j + R_{\text{veh}}) < 0] \Pi_{ib} \sin(\pi/2 * \lceil z_{ib} / \Pi_{ib} \rceil)$.

To drive errors z_{i1} and z_{i2} to converge under constraints (12), the surge and yaw guidance velocities α_{iu} and α_{iw} with (16)–(18), are presented as follows:

$$\alpha_{iu} = \frac{1}{d_i} \left\{ \frac{1}{\mathcal{F}_{i1}} (k_{iu}^g \xi_{i1} + \mathcal{F}_{i1}^r z_{i1}^r) + \kappa_{i1} (\lambda_{i1}^r - \lambda_{i1}^l) - \frac{p_{ie}^T}{\|p_{ie}\|} \left[\sum_{j \in \mathcal{N}_i^F} a_{ij} \mathcal{R}_j v_j + \sum_{k \in \mathcal{N}_i^L} a_{ik} p_{kd}^{\theta_k} v_s \right] + 2u_i \sin^2 \left(\frac{z_{i2}}{2} \right) + v_i \sin(z_{i2}) \right\} \quad (19)$$

$$\alpha_{iw} = -\frac{1}{\mathcal{F}_{i2}} \left(k_{iw}^g \xi_{i2} + \mathcal{F}_{i2}^r z_{i2}^r + \mathcal{F}_{i2}^l z_{i2}^l \right) + \dot{\psi}_{id} + \kappa_{i2} \left(\lambda_{i2}^r - \lambda_{i2}^l \right) \quad (20)$$

where $k_{iu}^g, k_{iw}^g \in \mathfrak{R}^+$. Note that it always holds $z_{i1} \geq 0$ by the definition of the relative distance in (11). Thus, z_{i1}^l is set to zero, and one has $z_{i1}^r = 0$.

To coordinate with ASVs and virtual leaders, the path variable θ_k for the k th virtual leader is updated by

$$\dot{\theta}_k = v_s - \varpi_k = v_s + \iota_k \phi_k \quad (21)$$

where $\iota_k \in \mathfrak{R}^+$ is a constant; $\phi_k = \phi_{k1} - \phi_{k2}$ with

$$\begin{cases} \phi_{k1} = \sum_{i \in \mathcal{N}_k^F} a_{ki} \mathcal{F}_{i1} \frac{p_{ie}^T}{\|p_{ie}\|} p_{kd}^{\theta_k} \xi_{i1} \\ \phi_{k2} = \sum_{l \in \mathcal{N}_k^L} a_{kl} \theta_{kl} + a_{k0} \theta_{ke} \end{cases} \quad (22)$$

where $\theta_{kl} = \theta_k - \theta_l - \mathcal{P}_{kl}$ and $\theta_{ke} = \theta_k - \theta_0 - \mathcal{P}_k$ with $\mathcal{P}_{kl} = \mathcal{P}_k - \mathcal{P}_l$ represent the coordination errors between virtual leaders and between virtual leaders and the super leader, respectively.

Combining (19), (20), and (22) with (18), (17), (16), and (21), the kinematic error dynamics can be given by

$$\begin{cases} \dot{\xi}_{i1} = -k_{iu}^g \xi_{i1} - \mathcal{F}_{i1} \left(d_i u_{ie} - \frac{p_{ie}^T}{\|p_{ie}\|} \sum_{k \in \mathcal{N}_i^L} a_{ik} p_{kd}^{\theta_k} \varpi_k \right) \\ \dot{\xi}_{i2} = -k_{iw}^g \xi_{i2} + \mathcal{F}_{i2} w_{ie} \\ \dot{\theta}_{ke} = -\varpi_k \end{cases} \quad (23)$$

where $u_{ie} = u_i - \alpha_{iu}$ and $w_{ie} = w_i - \alpha_{iw}$.

B. PTDO-Based Antidisturbance Control Laws

For underactuated ASVs, we only consider its surge and yaw dynamics expressed as follows:

$$\begin{cases} \dot{u}_i = \sigma_{iu}(u_i, v_i, w_i) + \frac{1}{m_{iu}} (\tau_{iu}^d + \tau_{iu}) \\ \dot{w}_i = \sigma_{iw}(u_i, v_i, w_i) + \frac{1}{m_{iw}} (\tau_{iw}^d + \tau_{iw}) \end{cases} \quad (24)$$

where $\sigma_{iu}(u_i, v_i, w_i) = (m_{iv} v_i w_i - d_{iu} u_i) / m_{iu}$ and $\sigma_{iw}(u_i, v_i, w_i) = [(m_{iu} - m_{iv}) u_i v_i - d_{iw} w_i] / m_{iw}$ are available functions.

Based on the simplified model (24), two auxiliary variables $\zeta_{iu} \in \mathfrak{R}$ and $\zeta_{iw} \in \mathfrak{R}$ are introduced to construct the reduced-order DO. To improve the estimation performance for external disturbances τ_{iu}^d and τ_{iw}^d , the following PTDOs with (2) are developed as follows:

$$\begin{cases} \hat{\tau}_{iu}^d = k_{iu}^{1o} \Gamma_{iu} + k_{iu}^{2o} \frac{\Gamma_{iu}}{|\Gamma_{iu}| + \Delta_{iu}} + k_{iu}^{3o} \frac{\dot{\mu}}{\mu} \Gamma_{iu} \\ \dot{\zeta}_{iu} = \sigma_{iu}(u_i, v_i, w_i) + \frac{1}{m_{iu}} (\hat{\tau}_{iu}^d + \tau_{iu}) \\ \hat{\tau}_{iw}^d = k_{iw}^{1o} \Gamma_{iw} + k_{iw}^{2o} \frac{\Gamma_{iw}}{|\Gamma_{iw}| + \Delta_{iw}} + k_{iw}^{3o} \frac{\dot{\mu}}{\mu} \Gamma_{iw} \\ \dot{\zeta}_{iw} = \sigma_{iw}(u_i, v_i, w_i) + \frac{1}{m_{iw}} (\hat{\tau}_{iw}^d + \tau_{iw}) \end{cases} \quad (25)$$

where $\hat{\tau}_{iu}^d \in \mathfrak{R}$ and $\hat{\tau}_{iw}^d \in \mathfrak{R}$ represent the estimation values of τ_{iu}^d and τ_{iw}^d , respectively; $\Gamma_{iu} = u_i - \zeta_{iu}$ and $\Gamma_{iw} = w_i - \zeta_{iw}$; $k_{iu}^{1o} \in \mathfrak{R}^+$, $k_{iu}^{2o} \in \mathfrak{R}^+$, $k_{iu}^{3o} \in \mathfrak{R}^+$, $k_{iw}^{1o} \in \mathfrak{R}^+$, $k_{iw}^{2o} \in \mathfrak{R}^+$, and $k_{iw}^{3o} \in \mathfrak{R}^+$ are observer gains; $\Delta_{iu} \in \mathfrak{R}^+$ and $\Delta_{iw} \in \mathfrak{R}^+$ are small scalars.

Define $z_{i3} = u_{ie}$ and $z_{i4} = w_i - \alpha_{iw}^*$ and take the derivatives of z_{i3} and z_{i4} along (24) as follows:

$$\begin{cases} \dot{z}_{i3} = \sigma_{iu}(u_i, v_i, w_i) + \frac{1}{m_{iu}} (\tau_{iu}^d + \tau_{iu}) - \dot{\alpha}_{iu} \\ \dot{z}_{i4} = \sigma_{iw}(u_i, v_i, w_i) + \frac{1}{m_{iw}} (\tau_{iw}^d + \tau_{iw}) - \dot{\alpha}_{iw}^* \end{cases} \quad (26)$$

With antidisturbance rejection control technique, the surge and yaw control laws based on PTDOs (25) are devised as follows:

$$\begin{cases} \tau_{iu} = m_{iu} [-k_{iu}^c z_{i3} - \sigma_{iu}(u_i, v_i, w_i) + \dot{\alpha}_{iu}] - \hat{\tau}_{iu}^d \\ \tau_{iw} = m_{iw} [-k_{iw}^c z_{i4} - \sigma_{iw}(u_i, v_i, w_i) + \dot{\alpha}_{iw}^*] - \hat{\tau}_{iw}^d \end{cases} \quad (27)$$

where $k_{iu}^c \in \mathfrak{R}^+$ and $k_{iw}^c \in \mathfrak{R}^+$ are control gain constants.

Substituting (27) into (26), the kinetic error dynamics are presented as follows:

$$\begin{cases} \dot{z}_{i3} = -k_{iu}^c z_{i3} + \frac{1}{m_{iu}} \tilde{\tau}_{iu}^d \\ \dot{z}_{i4} = -k_{iw}^c z_{i4} + \frac{1}{m_{iw}} \tilde{\tau}_{iw}^d \end{cases} \quad (28)$$

with $\tilde{\tau}_{iu}^d = \tau_{iu}^d - \hat{\tau}_{iu}^d$ and $\tilde{\tau}_{iw}^d = \tau_{iw}^d - \hat{\tau}_{iw}^d$.

C. Barrier-Certified Yaw Velocity Protocol

In the previous section, performance-prescribed surge and yaw guidance laws have been developed for underactuated ASVs. The proposed guidance laws (19) and (20) are viewed as nominal guidance laws for stabilizing the relative distance and heading errors, which cannot ensure the safety of multi-ASV system. Thus, this part will present a barrier-certified yaw velocity protocol to modify the nominal guidance laws for the safety of multi-ASV system.

Before designing the barrier-certified yaw velocity protocol, rewrite the kinematic dynamics of ASVs as follows:

$$\begin{bmatrix} \dot{p}_i \\ \dot{\phi}_i \end{bmatrix} = \underbrace{\begin{bmatrix} \mathcal{R}_i(\phi_i) v_i \\ 0 \end{bmatrix}}_{f_i} + \underbrace{\begin{bmatrix} 0 \\ 1 \end{bmatrix}}_{g_i} \times w_i. \quad (29)$$

In order to ensure no collision with the o th obstacle, the i th ASV is forced to leave obstacle avoidance zone described by Definition 1, i.e., $p_i \notin \mathcal{C}_{io}(p_i)$ for $\forall t \geq t_0$. Motivated by the set invariance [50], a set $\bar{\mathcal{C}}_{io}(p_i)$ for the i th ASV is defined as follows:

$$\bar{\mathcal{C}}_{io}(p_i) = \{p_i \in \mathfrak{R}^2 \mid h_{io}(p_i) \geq 0\} \quad (30)$$

where $\bar{\mathcal{C}}_{io}(p_i)$ is regarded as a complement of $\mathcal{C}_{io}(p_i)$; $h_{io}(p_i) = \|p_i - p_o\|^2 - (r_o + R_{\text{obs}})^2$ is a CBF.

Because the first-order derivative of h_{io} along system (29) does not contain the velocity signal w_i , h_{io} is regarded as a second-order CBF for system (29) by the definition of relative degree [50]. Next, the following sets $\bar{\mathcal{C}}_{io,1}(p_i)$, and $\bar{\mathcal{C}}_{io,2}(p_i)$ are presented as:

$$\begin{cases} \bar{\mathcal{C}}_{io,1}(p_i) = \{p_i \in \mathfrak{R}^2 \mid \chi_{io,0}(p_i) \geq 0\} \\ \bar{\mathcal{C}}_{io,2}(p_i) = \{p_i \in \mathfrak{R}^2 \mid \chi_{io,1}(p_i) \geq 0\} \end{cases} \quad (31)$$

where $\chi_{io,0}$ and $\chi_{io,1}$ are the differentiable functions expressed by $\chi_{io,0}(p_i) = h_{io}(p_i)$ and $\chi_{io,1}(p_i) = \dot{\chi}_{io,0}(p_i) + \beta_{io,1}\chi_{io,0}(p_i)$ with $\beta_{io,1} \in \mathfrak{R}^+$. Then, the forward invariance of $\bar{C}_{io,1}(p_i)$ can be guaranteed if the set $\bar{C}_{io,1}(p_i) \cap \bar{C}_{io,2}(p_i)$ is forward invariant [50].

To ensure the forward invariance of the set $\bar{C}_{io,1}(p_i) \cap \bar{C}_{io,2}(p_i)$, an obstacle-avoided yaw velocity constraint is yielded as follows:

$$\mathcal{W}_{io} = \left\{ w_i \in \mathfrak{R} \mid L_{f_i}^2 h_{io} + L_{g_i} L_{f_i} h_{io} w_i + O_{io}(h_{io}) + \beta_{io,2} \chi_{io,1} \geq 0 \right\} \quad (32)$$

where $\beta_{io,2} \in \mathfrak{R}^+$; $L_{f_i}^2 h_{io}$ and $L_{g_i} L_{f_i} h_{io}$ are Lie derivatives for system (29); $O_{io}(h_{io})$ denotes the remaining Lie derivatives with the relative degree 0 or 1.

According to Definition 2, a set $\bar{S}_{ij}(p_i)$, viewed as a complement of $S_{ij}(p_i)$, is defined to describe the safety objective among ASVs as follows:

$$\bar{S}_{ij}(p_i) = \{p_i \in \mathfrak{R}^2 \mid h_{ij}(p_i) \geq 0\} \quad (33)$$

with $h_{ij}(p_i) = \|p_i - p_j\|^2 - (r_i + r_j + R_{\text{veh}})^2$. Similar to (31), two sets associated with $\bar{S}_{ij}(p_i)$ are presented as follows:

$$\begin{aligned} \bar{S}_{ij,1}(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{ij,0}(p_i) \geq 0\} \\ \bar{S}_{ij,2}(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{ij,1}(p_i) \geq 0\} \end{aligned} \quad (34)$$

where $\chi_{ij,0}(p_i) = h_{ij}(p_i)$ and $\chi_{ij,1}(p_i) = \dot{\chi}_{ij,0}(p_i) + \beta_{ij,1}\chi_{ij,0}(p_i)$ with $\beta_{ij,1} \in \mathfrak{R}^+$.

Then, an ASV-avoided yaw velocity constraint is given by

$$\mathcal{W}_{ij} = \left\{ w_i \in \mathfrak{R} \mid L_{f_i}^2 h_{ij} + L_{g_i} L_{f_i} h_{ij} w_i + O_{ij}(h_{ij}) + \beta_{ij,2} \chi_{ij,1} \geq 0 \right\} \quad (35)$$

where $\beta_{ij,2} \in \mathfrak{R}^+$; $L_{f_i}^2 h_{ij}$ and $L_{g_i} L_{f_i} h_{ij}$ are Lie derivatives; and $O_{ij}(h_{ij})$ is also the remaining Lie derivative.

The barrier-certified yaw velocity for the safety of the i th ASV must satisfy the constraint $\mathcal{U}_{io} \cap \mathcal{U}_{ij}$, $o \in \{1, \dots, N_o\}$, $j \in \{1, \dots, M\} \setminus \{i\}$. Thus, a barrier-certified yaw velocity protocol is formulated by the following quadratic optimization:

$$\begin{aligned} \alpha_{iw}^* &= \arg \min_{w_i \in \mathfrak{R}} J(w_i) = \|w_i - \alpha_{iw}\|^2 \\ \text{s.t.} \quad w_i &\in \mathcal{W}_{io} \cap \mathcal{W}_{ij}, \quad o \in \{1, \dots, N_o\}, j \in \mathcal{V}^F \setminus \{i\} \end{aligned} \quad (36)$$

where α_{iw}^* is the optimal yaw velocity for stability and safety.

To facilitate the implementation of the optimization problem (36), an online optimization technique is employed by an recurrent neural network (RNN)

$$\dot{w}_i = -\frac{1}{\epsilon_i} \left[\nabla J(w_i) + \varsigma_i \sum_{l=1}^{M+N_o-1} \partial \max\{0, \Xi_{il}(w_i)\} \right] \quad (37)$$

where $\epsilon_i, \varsigma_i \in \mathfrak{R}^+$; $\Xi_{il}(w_i) = -L_{f_i}^2 h_{ij} - L_{g_i} L_{f_i} h_{ij} w_i - O_{ij}(h_{ij}) - \beta_{ij,2}(h_{ij})$, $l = 1, \dots, M-1$, $\Xi_{il}(w_i) = -L_{f_i}^2 h_{io} - L_{g_i} L_{f_i} h_{io} w_i -$

$O_{io}(h_{io}) - \beta_{io,2}(h_{io})$, $l = M, \dots, M+N_o-1$; $\partial \max\{0, \Xi_{il}(w_i)\}$ represents a piece-wise penalty function defined as follows:

$$\partial \max\{0, \Xi_{il}(w_i)\} = \begin{cases} \nabla \Xi_{il}(w_i), & \Xi_{il}(w_i) > 0 \\ [0, \nabla \Xi_{il}(w_i)], & \Xi_{il}(w_i) = 0 \\ 0, & \Xi_{il}(w_i) < 0. \end{cases}$$

According to [53], the neuronal state w_i of RNN (37) can converge to the optimal solution α_{iw}^* within a finite time.

IV. MAIN RESULTS

In the previous section, constrained safe cooperative maneuvering method has been proposed for multiple ASVs in an obstacle-loaded environment. The section analyzes the stability and the safety of the closed-loop system.

A. Observation Subsystem

Lemma 2: Under Assumption 2, when $k_{iu}^{2o} \geq \bar{\tau}_i^d$ and $k_{iw}^{2o} \geq \bar{\tau}_i^d$, the disturbance terms τ_{iu}^d and τ_{iw}^d can be precisely estimated by using proposed PTDOs (25) within a prescribed time. Estimation errors $\tilde{\tau}_{iu}^d$ and $\tilde{\tau}_{iw}^d$ are prescribed-time stable.

Proof: Consider a Lyapunov function candidate as follows:

$$V_{i1} = \frac{1}{2} (m_{iu} \Gamma_{iu}^2 + m_{iw} \Gamma_{iw}^2). \quad (38)$$

With (24) and (25), it takes the derivative of V_{i1} as follows:

$$\begin{aligned} \dot{V}_{i1} &= m_{iu} \Gamma_{iu} (\dot{u}_i - \dot{\zeta}_{iu}) + m_{iw} \Gamma_{iw} (\dot{w}_i - \dot{\zeta}_{iw}) \\ &= -\Gamma_{iu} \left(k_{iu}^{1o} \Gamma_{iu} + k_{iu}^{2o} \frac{\Gamma_{iu}}{|\Gamma_{iu}| + \Delta_{iu}} + k_{iu}^{3o} \frac{\dot{\mu}}{\mu} \Gamma_{iu} - \tau_{iu}^d \right) \\ &\quad - \Gamma_{iw} \left(k_{iw}^{1o} \Gamma_{iw} + k_{iw}^{2o} \frac{\Gamma_{iw}}{|\Gamma_{iw}| + \Delta_{iw}} + k_{iw}^{3o} \frac{\dot{\mu}}{\mu} \Gamma_{iw} - \tau_{iw}^d \right). \end{aligned}$$

Let $\Gamma_i = [\Gamma_{iu}, \Gamma_{iw}]^T$, $K_i^{1o} = \text{diag}\{k_{iu}^{1o}, k_{iw}^{1o}\}$, $K_i^{2o} = \text{diag}\{k_{iu}^{2o}/(|\Gamma_{iu}| + \Delta_{iu}), k_{iw}^{2o}/(|\Gamma_{iw}| + \Delta_{iw})\}$, and $K_i^{3o} = \text{diag}\{k_{iu}^{3o}, k_{iw}^{3o}\}$. Under Assumption 2, we have

$$\begin{aligned} \dot{V}_{i1} &\leq -\underline{\lambda}(K_i^{1o}) \|\Gamma_i\|^2 - \underline{\lambda}(K_i^{2o}) \|\Gamma_i\| \\ &\quad - \underline{\lambda}(K_i^{3o}) \frac{\dot{\mu}}{\mu} \|\Gamma_i\|^2 + \bar{\tau}_{id} \|\Gamma_i\|. \end{aligned} \quad (39)$$

When $\underline{\lambda}(K_i^{2o}) \geq \bar{\tau}_{id}$, it renders that

$$\dot{V}_{i1} \leq -\underline{\lambda}(K_i^{1o}) \|\Gamma_i\|^2 - \underline{\lambda}(K_i^{3o}) \frac{\dot{\mu}}{\mu} \|\Gamma_i\|^2. \quad (40)$$

Since $\underline{\lambda}(K_i^{3o}) \geq \bar{m}_i$ with $\bar{m}_i = \max\{m_{iu}, m_{iw}\}$, it yields from (38) that

$$\begin{aligned} \dot{V}_{i1} &\leq -2 \frac{\underline{\lambda}(K_i^{1o})}{\bar{m}_i} V_{i1} - 2 \frac{\underline{\lambda}(K_i^{3o})}{\bar{m}_i} \frac{\dot{\mu}}{\mu} V_{i1} \\ &\leq -2 \frac{\underline{\lambda}(K_i^{1o})}{\bar{m}_i} V_{i1} - 2 \frac{\dot{\mu}}{\mu} V_{i1}. \end{aligned} \quad (41)$$

According to Lemma 1, it obtains

$$V_{i1}(t) \leq \mu^{-2}(t) e^{-2 \frac{\underline{\lambda}(K_i^{1o})}{\bar{m}_i} (t-t_0)} V_{i1}(t_0) \quad (42)$$

on $t \in [t_0, T_1]$. It also implies that $\|\Gamma_i(t)\| \leq \mu^{-2} e^{-2 \frac{\underline{\lambda}(K_i^{1o})}{\bar{m}_i} (t-t_0)} \|\Gamma_i(t_0)\|$ on $t \in [t_0, T_1]$. Further, it renders that $\lim_{t \rightarrow T_1^-} \|\Gamma_i(t)\| \mapsto 0$. Then, we have $\dot{V}_{i1}(t) = 0$ for

$t \in [T_1, \infty)$ and $\dot{\Gamma} = 0$ for $t \in [T_1, \infty)$. Using (24) and (25), the estimation error always holds $\tilde{\tau}_i^d = [\tilde{\tau}_{iu}^d, \tilde{\tau}_{iw}^d]^T = \bar{M}_i \bar{\Gamma}_i$. Hence, it infers that $\tilde{\tau}_i^d = 0$ for $t \in [T_1, \infty)$. ■

B. Control Subsystems

Lemma 3: The kinetic subsystem (28) with states z_{i3} and z_{i4} and inputs $\tilde{\tau}_{iu}^d$ and $\tilde{\tau}_{iw}^d$ is input-to-state stable.

Proof: Consider a Lyapunov function candidate

$$V_{i2} = \sqrt{\frac{1}{2}(z_{i3}^2 + z_{i4}^2)} \quad (43)$$

and take its derivative along (28) as follows:

$$\begin{aligned} \dot{V}_{i2} &= \frac{1}{2V_{i2}}(z_{i3}\dot{z}_{i3} + z_{i4}\dot{z}_{i4}) \\ &= -\frac{1}{2V_{i2}}(k_{iu}^c z_{i3}^2 + k_{iw}^c z_{i4}^2) \\ &\quad + \frac{1}{2V_{i2}}\left(\frac{1}{m_{iu}}z_{i3}\tilde{\tau}_{iu}^d + \frac{1}{m_{iw}}z_{i4}\tilde{\tau}_{iw}^d\right) \\ &\leq -k_i^c V_{i2} + \frac{1}{\sqrt{2m_i}}\|\tilde{\tau}_i^d\| \end{aligned} \quad (44)$$

where $k_i^c = \min\{k_{iu}^c, k_{iw}^c\}$ and $m_i = \min\{m_{iu}, m_{iw}\}$. From Lemma 2, $\|\tilde{\tau}_i^d\|$ is bounded such that the kinetic subsystem (28) is input-to-state stable. ■

Lemma 4: The kinematic subsystem (23) with states ξ_{i1} , ξ_{i2} , and θ_{ke} and inputs u_{ie} and w_{ie} is input-to-state stable.

Proof: Define vectors $\theta_e = [\theta_{(M+1)e}, \dots, \theta_{Ne}]^T$, $\phi_1 = [\phi_{(M+1)1}, \dots, \phi_{N1}]^T$, and $\phi_2 = [\phi_{(M+1)2}, \dots, \phi_{N2}]^T$. Then, it follows from (22) that:

$$\phi_2 = \mathcal{H}\theta_e \quad (45)$$

with $\mathcal{H} = \mathcal{L}_0 + \mathcal{B}_0$, where \mathcal{B}_0 is a diagonal matrix with diagonal element being 1 only if the super leader's information is available.

Consider a Lyapunov function candidate V_3 as follows:

$$V_3 = \sum_{i=1}^M \frac{1}{2}(\xi_{i1}^2 + \xi_{i2}^2) + \frac{1}{2}\theta_e^T \mathcal{H}\theta_e. \quad (46)$$

Using (45), it takes the derivative of V_3 as follows:

$$\dot{V}_3 = \sum_{i=1}^M (\xi_{i1}\dot{\xi}_{i1} + \xi_{i2}\dot{\xi}_{i2}) + \sum_{k=M+1}^N \phi_{k2}^T \dot{\theta}_{ke}. \quad (47)$$

Along dynamics (22) and (23), it yields that

$$\begin{aligned} \dot{V}_3 &= \sum_{i=1}^M \left(-k_{iu}^g \xi_{i1}^2 - k_{iw}^g \xi_{i2}^2 - d_i \mathcal{F}_{i1} \xi_{i1} u_{ie} + \mathcal{F}_{i2} \xi_{i2} w_{ie} \right. \\ &\quad \left. + \mathcal{F}_{i1} \xi_{i1} \frac{p_{ie}^T}{\|p_{ie}\|} \sum_{k \in \mathcal{N}_i^T} a_{ik} p_{kd}^{\theta_k} \varpi_k \right) - \sum_{k=M+1}^N (\phi_{k1}^T - \phi_k^T) \varpi_k \\ &= \sum_{i=1}^M \left(-k_{iu}^g \xi_{i1}^2 - k_{iw}^g \xi_{i2}^2 - d_i \xi_{i1} \mathcal{F}_{i1} u_{ie} + \xi_{i2} \mathcal{F}_{i2} w_{ie} \right) \\ &\quad - \sum_{k=M+1}^N \iota_k \phi_k^2. \end{aligned} \quad (48)$$

Letting $\xi_i = [\xi_{i1}, \xi_{i2}]^T$ and $\vartheta_{ie} = [u_{ie}, w_{ie}]^T$, it renders from (48) that

$$\begin{aligned} \dot{V}_3 &\leq \sum_{i=1}^M \left(-\underline{\lambda}(K_i^g) \|\xi_i\|^2 + d_i \bar{\lambda}(\mathcal{F}_i) \|\xi_i\| \|\vartheta_{ie}\| \right) \\ &\quad - \sum_{k=M+1}^N \iota_k \phi_k^2 \end{aligned} \quad (49)$$

where $K_i^g = \text{diag}\{k_{iu}^g, k_{iw}^g\}$ and $\mathcal{F}_i = \text{diag}\{\mathcal{F}_{i1}, \mathcal{F}_{i2}\}$.

Define $E' = [\xi^T, \phi^T]^T$ and $E = [\xi^T, \theta_e^T]^T$ with $\xi = [\xi_1^T, \dots, \xi_M^T]^T$ and $\phi = [\phi_{M+1}, \dots, \phi_N]^T$. According to the fact that $E' = \Lambda E$ where

$$\Lambda = \begin{bmatrix} I_M \otimes I_2 & 0_{2M \times (N-M)} \\ \Psi & -\mathcal{H} \end{bmatrix}$$

$$\Psi = \begin{bmatrix} \Psi_{(M+1)1} & 0 & \cdots & \Psi_{(M+1)M} & 0 \\ \vdots & \vdots & \ddots & \vdots & \vdots \\ \Psi_{N1} & 0 & \cdots & \Psi_{NM} & 0 \end{bmatrix}$$

with $\Psi_{ki} = a_{ki} \mathcal{F}_{i1} p_{ie}^T p_{kd}^{\theta_k} / \|p_{ie}\|$, $i = 1, \dots, M$, $k = M+1, \dots, N$, it further follows that:

$$\begin{aligned} \dot{V}_3 &\leq -c \|E'\|^2 + \bar{\lambda}_{\mathcal{F}} \|E\| \|d\| \|\vartheta_e\| \\ &\leq -c \underline{\lambda}(\Lambda) \|E\|^2 + \bar{\lambda}_{\mathcal{F}} \|E\| \|d\| \|\vartheta_e\| \end{aligned} \quad (50)$$

where $c = \min_{i=1, \dots, M, k=M+1, \dots, N} \{\underline{\lambda}(K_i^g), \iota_k\}$, $\bar{\lambda}_{\mathcal{F}} = \max_{i=1, \dots, M} \{\bar{\lambda}(\mathcal{F}_i)\}$, $d = \text{diag}\{d_1, 0, \dots, d_M, 0\}$, and $\vartheta_e = [\vartheta_{1e}^T, \dots, \vartheta_{Me}^T]^T$.

For $\|E\| \geq \bar{\lambda}_{\mathcal{F}} \|d\| \|\vartheta_e\| / (bc \underline{\lambda}(\Lambda))$ with $b \in (0, 1)$, we have

$$\dot{V}_3 \leq -c \underline{\lambda}(\Lambda) (1-b) \|E\|^2. \quad (51)$$

Note that $u_{ie} = z_{i3}$ and $w_{ie} = z_{i4} + \alpha_{iw}^* - \alpha_{iw}$, it gets that u_{ie} and w_{ie} are bounded from Lemma 3. Further, it is deduced that ϑ_e is bounded satisfying $\|\vartheta_e\| \leq \bar{\vartheta}_e \in \mathfrak{R}^+$. Consequently, it concludes that the subsystem (23) is input-to-state stable. ■

C. Optimization Subsystem

Lemma 5: For the underactuated ASVs with dynamic (3) and safe velocity constraints (32) and (35), the sets $\bar{\mathcal{C}}_{io}(p_i)$ and $\bar{\mathcal{S}}_{ij}(p_i)$ are guaranteed to be input-to-state safe for $p_i(t_0) \in \bar{\mathcal{C}}_{io}(p_i)$ and $p_i(t_0) \in \bar{\mathcal{S}}_{ij}(p_i)$, i.e., $p_i(t) \in \bar{\mathcal{C}}_{io}(p_i)$ and $p_i(t) \in \bar{\mathcal{S}}_{ij}(p_i) \forall t \geq t_0$.

Proof: From Lemma 3, it is found that w_i cannot asymptotically converge to α_{iw}^* affected by the bounded disturbance $\tilde{\tau}_{iw}^d$. Thus, it ensure that $\bar{\mathcal{C}}_{io}(p_i)$ and $\bar{\mathcal{S}}_{ij}(p_i)$ are input-to-state safe in the presence of the bounded disturbance $\tilde{\tau}_{iw}^d$. The slightly larger sets associated with $\tilde{\tau}_{iw}^d$ are defined as follows:

$$\begin{aligned} \bar{\mathcal{C}}_{io,1}^d(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{io,0}(p_i) + \varrho_{io,1}(\|\tilde{\tau}_{iw}^d\|_\infty) \geq 0\} \\ \bar{\mathcal{C}}_{io,2}^d(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{io,1}(p_i) + \varrho_{io,2}(\|\tilde{\tau}_{iw}^d\|_\infty) \geq 0\} \\ \bar{\mathcal{S}}_{ij,1}^d(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{ij,0}(p_i) + \varrho_{ij,1}(\|\tilde{\tau}_{iw}^d\|_\infty) \geq 0\} \\ \bar{\mathcal{S}}_{ij,2}^d(p_i) &= \{p_i \in \mathfrak{R}^2 \mid \chi_{ij,1}(p_i) + \varrho_{ij,2}(\|\tilde{\tau}_{iw}^d\|_\infty) \geq 0\} \end{aligned} \quad (52)$$

where $\varrho_{io,1}(\cdot)$, $\varrho_{io,2}(\cdot)$, $\varrho_{ij,1}(\cdot)$, and $\varrho_{ij,2}(\cdot)$ are class \mathcal{K} functions.

By $\bar{C}_{io,2}^d(p_i)$ and $\bar{S}_{ij,2}^d(p_i)$, an extended set is expressed by

$$\Omega_i(p_i) = \{p_i \in \mathfrak{R}^2 \mid \bar{\chi}_{io,1}(p_i) \geq 0, \bar{\chi}_{ij,1}(p_i) \geq 0\} \quad (53)$$

where $\bar{\chi}_{io,1}(p_i) = -V_{i2} + \ell_{io}\chi_{io,1}(p_i) + \ell_{io}q_{io,2}(\|\bar{\tau}_i^d\|_\infty)$ and $\bar{\chi}_{ij,1}(p_i) = -V_{i2} + \ell_{ij}\chi_{ij,1}(p_i) + \ell_{ij}q_{ij,2}(\|\bar{\tau}_i^d\|_\infty)$ with $\ell_{io} \in \mathfrak{R}^+$ and $\ell_{ij} \in \mathfrak{R}^+$.

Using $\ell_{i\star}\beta_{i\star,2}q_{i\star,2}(\|\bar{\tau}_i^d\|_\infty) = \|\bar{\tau}_i^d\|/\sqrt{2}m_i$ with $\star = o, j$, the derivative of $\bar{\chi}_{i\star,1}$ is put into

$$\begin{aligned} \dot{\bar{\chi}}_{i\star,1} &= -\dot{V}_{i2} + \ell_{i\star}(L_{f_i}\bar{\chi}_{i\star,1} + L_{g_i}\bar{\chi}_{i\star,1}w_i) \\ &= -\dot{V}_{i2} + \ell_{i\star}(L_{f_i}\bar{\chi}_{i\star,1} + L_{g_i}\bar{\chi}_{i\star,1}(\alpha_{i\star}^* + z_{i4})) \\ &\geq k_i^c V_{i2} - \frac{1}{\sqrt{2}m_i} \|\bar{\tau}_i^d\| - \ell_{i\star}\beta_{i\star,2}\chi_{i\star,1} - \ell_{i\star}|L_{g_i}\bar{\chi}_{i\star,1}| |z_{i4}| \\ &\geq (k_i^c - \beta_{i\star,2})V_{i2} - \beta_{i\star,2}\bar{\chi}_{i\star,1} - \ell_{i\star}|L_{g_i}\bar{\chi}_{i\star,1}| |z_{i4}| \end{aligned} \quad (54)$$

for $p_i(t_0) \in \Omega_i(p_i)$.

When $k_i^c > \beta_{i\star,2}$, it follows that $\dot{\bar{\chi}}_{i\star,1} \geq -\beta_{i\star,2}\bar{\chi}_{i\star,1}$. According to [51], the set $\Omega_i(p_i)$ for $p_i(t_0) \in \Omega_i(p_i)$ is forward invariant, i.e., $p_i(t) \in \Omega_i(p_i) \quad \forall t > t_0$. Obviously, it gets that $\chi_{i\star,1}(p_i) + q_{i\star,2}(\|\bar{\tau}_i^d\|_\infty) \geq V_{i2}/\ell_{i\star} \geq 0$. Thus, these sets $\bar{C}_{io,2}^d(p_i)$ and $\bar{S}_{ij,2}^d(p_i)$ are forward invariant. According to the proof of [52, Lemma 3], it renders that $\bar{C}_{io,1}^d(p_i)$ and $\bar{S}_{ij,1}^d(p_i)$ are also forward invariant, i.e., $\bar{C}_{io,1}^d(p_i)$ and $\bar{S}_{ij,1}^d(p_i)$ are safe. Since $\bar{C}_{io,1}^d(p_i) \supset \bar{C}_{io,1}(p_i)$ and $\bar{S}_{ij,1}^d(p_i) \supset \bar{S}_{ij,1}(p_i)$ from (31), (34) and (52), $\bar{C}_{io,1}(p_i)$ and $\bar{S}_{ij,1}(p_i)$ are input-to-state safe. By the fact that $\chi_{io,0} = h_{io}$ and $\chi_{ij,0} = h_{ij}$, it obtains $\bar{C}_{io,1}^d(p_i) = \bar{C}_{io}(p_i)$ and $\bar{S}_{ij,1}^d(p_i) = \bar{S}_{ij}(p_i)$ such that $\bar{C}_{io}(p_i)$ and $\bar{S}_{ij}(p_i)$ are input-to-state safe. ■

D. Stability and Safety Results

It can be observed from (44) that $\bar{\tau}_{iu}^d$ and $\bar{\tau}_{iw}^d$ are inputs of the kinetic subsystem (28). It can be observed from (50) that the states of kinetic subsystem (28) z_{i3} and z_{i4} are some inputs of kinematic subsystem (23). The subsystem (25) and the system cascaded by subsystems (28) and (23) lead to the resulting closed-loop system. Lemmas 2–4 state the stability of all subsystems (23), (25), and (28). Lemma 5 gives the input-to-state safety of multi-ASV system. The stability and safety of the closed-loop system are given via the following theorem.

Theorem 1: Consider a swarm of underactuated ASVs expressed as dynamics (3) with the auxiliary system (18), the kinematic guidance laws (19) and (20), the update law (21), the barrier-certified yaw velocity (37), the PTDOs (25), and the kinetic control laws (27). Under Assumptions 1 and 2, it is ensured that: 1) all error signals of the closed-loop constrained safe cooperative maneuvering system are bounded; 2) the multi-ASV system is ensured to be input-to-state safe; and 3) the ATPP constraints (12) are not violated.

Proof: From [54, Lemmas 2–4 and 4.6], it has concluded that all error signals are bounded. According to definition (9), we have $p_e = \mathcal{L}_1 \otimes I_2[p + (\mathcal{L}_1^{-1}\mathcal{L}_2 \otimes I_2)p_d]$ with $p = [p_1^T, \dots, p_M^T]^T$ and $p_d = [p_{(M+1)d}^T, \dots, p_{Nd}^T]^T$. Further, it yields $\|p + (\mathcal{L}_1^{-1}\mathcal{L}_2 \otimes I_2)p_d\| \leq \|p_e\|/\underline{\lambda}(\mathcal{L}_1)$. Under Assumption 1, it gets that the error ξ is bounded by Lemma 4, which means that p_e is bounded by (11). Then, there exists a positive constant ε_{ip}

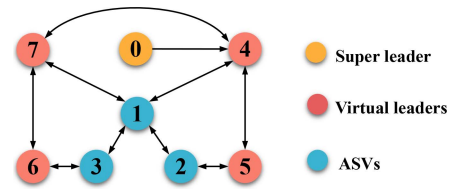


Fig. 2. Network topology.

such that the geometric objective (5) is satisfied. By Lemma 4 and (22), it yields that θ_e and ϕ_k are bounded. It implies that there exist two positive constants $\varepsilon_{k\theta 1}$ and $\varepsilon_{k\theta 2}$ such that the dynamic objective (6) is achieved. From Lemma 5, it has $p_i(t) \in \bar{C}_{io}(p_i)$ and $p_i(t) \in \bar{S}_{ij}(p_i) \quad \forall t \geq t_0$. According to the definitions of $\bar{C}_{io}(p_i)$ and $\bar{S}_{ij}(p_i)$, it gets that $p_i(t) \notin \bar{C}_{io}(p_i)$ and $p_i(t) \notin \bar{S}_{ij}(p_i) \quad \forall t \geq t_0$, i.e., safety objectives (7) and (8) are achieved. ■

V. SIMULATION RESULTS

This section conducts simulation results to verify the effectiveness of the proposed constrained safe cooperative maneuvering method. We consider three ASVs (labeled as ASV1, ..., ASV3), four virtual leaders, and one super leader. A network topology in Fig. 2 is given to formulate the cooperative maneuvering pattern. In addition, three obstacles, i.e., two stationary circular obstacles and one moving ship, are placed to evaluate the ability of collision avoidance.

Suppose that four virtual leaders move along four parametrized paths given as $p_{kd}(\theta_k) = [(\sqrt{2}/2)v_s\theta_k + (\sqrt{2}/2)(15 \cos([1/30]v_s\theta_k + [\pi/30]) + 40), (\sqrt{2}/2)v_s\theta_k - (\sqrt{2}/2)(15 \cos([1/30]v_s\theta_k + [\pi/30]) + 40)]^T$ for $k = 4, 5$, $p_{kd}(\theta_k) = [(\sqrt{2}/2)v_s\theta_k - (\sqrt{2}/2)(15 \cos([1/30]v_s\theta_k + [\pi/30]) + 40), [\sqrt{2}/2]v_s\theta_k + [\sqrt{2}/2](15 \cos([1/30]v_s\theta_k + [\pi/30]) + 40)]^T$ for $k = 6, 7$ with $v_s = 0.5$ and $\theta_4(0) = \theta_5(0) = \theta_6(0) = \theta_7(0) = 0$. The deviations of path parameters are set as $\mathcal{P}_4 = \mathcal{P}_7 = 0$ and $\mathcal{P}_5 = -\mathcal{P}_6 = -40$. For the vehicle model, we consider the ASV1–ASV3 with a length of 1.255 m, and the other Bis-scale parameters: $m_{iu} = 23.8$ kg, $m_{iv} = 33.8$ kg, $m_{iw} = 2.764$ kg, $d_{iu} = 2$, $d_{iv} = 7$, and $d_{iw} = 0.5$. According to Remark 1, the edge-nonsmooth obstacle can be modeled as a circular obstacle. For simplicity, we consider two static circular obstacles (named by Obstacle1 and Obstacle2), and an intersection ASV (named by Obstacle3). The parameters of all ASVs and obstacles are summarized into Table I, including the initial position, velocity, and radius. To conduct the simulation, the main equations and parameters of presented method are given as below: $\rho_{i1,0} = 10$; $\rho_{i2,0} = 2.0$; $\rho_{i1,\infty} = 1$; $\rho_{i2,\infty} = 0.4$; $\iota_{i1} = 0.05$; $\iota_{i2} = 0.2$; $\delta_{i1}^l = 0.7$; $\delta_{i1}^r = 0.4$; $\delta_{i2}^l = \delta_{i2}^r = 0.5$; $\kappa_{i1} = 0.5$; $\kappa_{i2} = 0.25$; $\Pi_{i1} = 10$; $\Pi_{i2} = \pi$; $k_{iu}^s = 0.5$; $k_{iw}^s = 1.0$; $\iota_k = 0.2$; $\beta_{io,1} = \beta_{io,2} = \beta_{ij,1} = \beta_{ij,2} = 0.5$; $R_{obs} = 2$; $R_{veh} = 1$; $\epsilon_i = 1$; $\zeta_i = 2$; $k_{iu}^{1o} = 0.5$; $k_{iu}^{2o} = 0.5$; $k_{iu}^{3o} = 0.5$; $k_{iw}^{1o} = 0.5$; $k_{iw}^{2o} = 0.5$; $k_{iw}^{3o} = 0.5$; $T_{iu} = 5$; $T_{iw} = 5$; $\Delta_{iu} = 1$; $\Delta_{iw} = 1$; $k_{iu}^c = 10$; and $k_{iw}^c = 10$.

Conducted by topology in Fig. 2, simulation results are plotted in Figs. 3–10, where gray bars in Figs. 5–10 represent the collision avoidance process of each ASV. Specifically, Fig. 3

TABLE I
 INITIAL VALUES OF ASVs AND OBSTACLES

$i, o = 1, 2, 3$	ASV1	ASV2	ASV3	Obstacle1	Obstacle2	Obstacle3
$[p_i^T(t_0), \varphi_i(t_0)]$ or $p_o^T(t_0)$	$[-2, 0, 0]$	$[15, -20, 0]$	$[-20, 12, 0]$	$[20, 35]$	$[82, 70]$	$[50, 120]$
$[\nu_i^T(t_0), w_i(t_0)]$ or $\dot{p}_o^T(t_0)$	$[0, 0, 0]$	$[0, 0, 0]$	$[0, 0, 0]$	$[0, 0]$	$[0, 0]$	$[0.1, 0]$
r_i or r_o	2	2	2	5	8	2

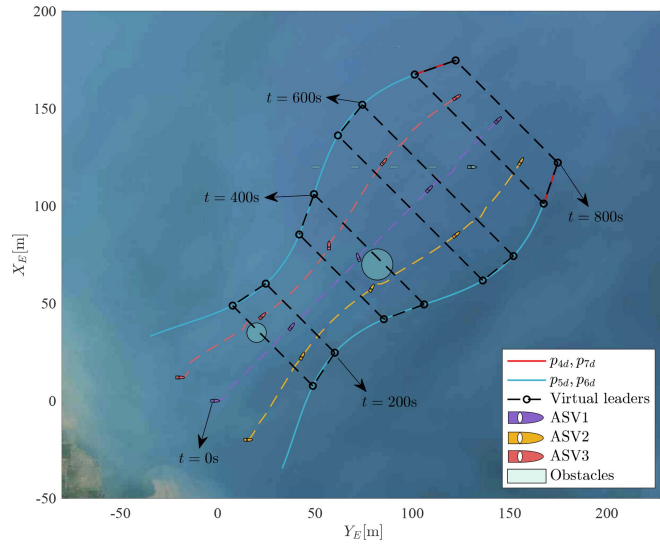


Fig. 3. Constrained safe cooperative maneuvering of three underactuated ASVs.

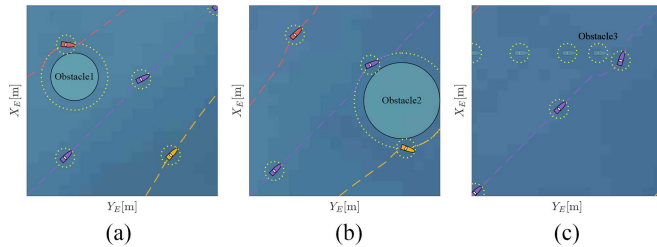


Fig. 4. Snapshots of collision avoidance for each obstacle. (a) Avoid Obstacle1. (b) Avoid Obstacle2. (c) Avoid Obstacle3.

shows the whole motion trajectories of ASV1–ASV3 guided by four virtual leaders. According to the snapshots at 0, 200, 400, 600, and 800 s, ASV1–ASV3 converge to the hulls and hold the collision-free cooperative maneuvering pattern. In order to clearly observe the collision avoidance process, Fig. 4 displays the enlarged snapshots, in which ASVs avoid Obstacle1, Obstacle2, and Obstacle3. From Fig. 4(a) and (b), each ASVs are capable of avoiding the static obstacles (Obstacle1 and Obstacle2). From Fig. 4(c), ASV1 can also avoid the intersection ASV (Obstacle3). The collision-free behaviors in Fig. 4 are further verified by the corresponding 2nd-order CBFs for each ASVs in Fig. 5. The non-negativity of $h_{io}(p_i)$ mean that each ASV does not go into the obstacle avoidance zones $\mathcal{C}_{io}(p_i)$, i.e., $p_i \notin \mathcal{C}_{io}(p_i)$. Then, it is concluded that the modified yaw guidance law (36) ensures the collision-free behaviors and achieve the cooperation of ASV1–ASV3 from Figs. 3 and 4.

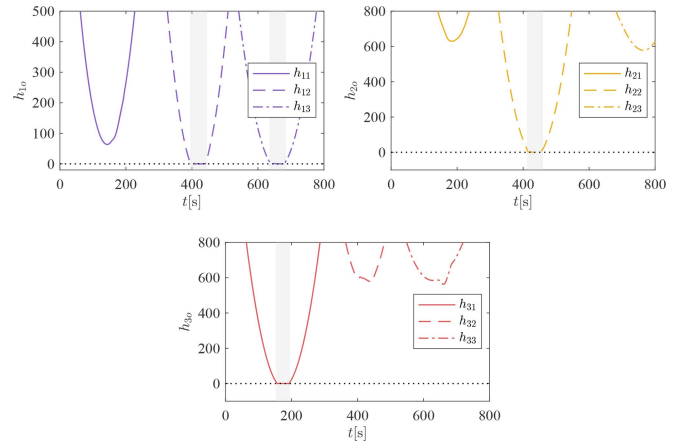


Fig. 5. Second-order CBFs of ASV1–ASV3 for obstacles.

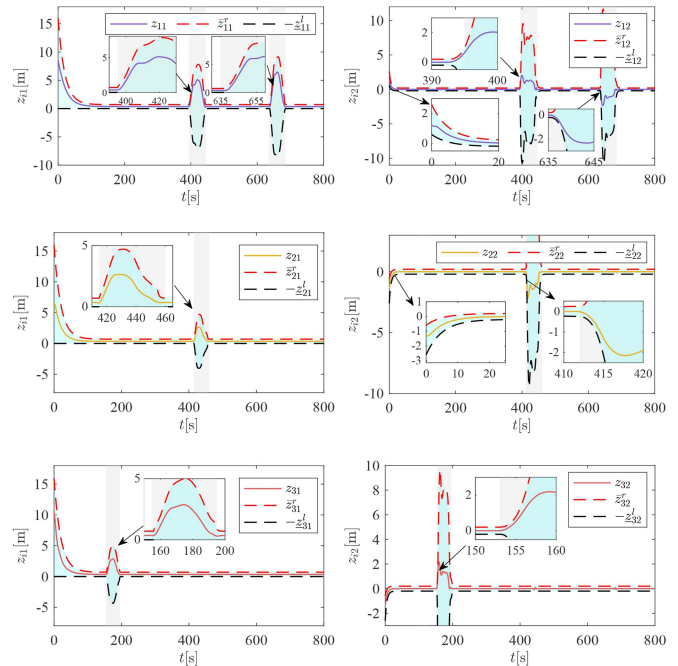


Fig. 6. Relative distances (left column) and heading errors (right column) of all ASVs.

From Fig. 6, the relative distances and heading errors of ASV1–ASV3 can converge into the prescribed constraint spaces $[-z_{i1}^l, \bar{z}_{i1}^r]$ and $[-z_{i2}^l, \bar{z}_{i2}^r]$. Under the collision-free protocol, z_{i1} escapes from the original constraint space $[-z_{i1}^l, \bar{z}_{i1}^r]$. To not violate the performance constraints, the space $[-z_{i1}^l, \bar{z}_{i1}^r]$ are enlarged to $[-z_{i1}^l, \bar{z}_{i1}^r]$ with the positive modified signals $\lambda_{i1}^l, \lambda_{i1}^r$ from proposed auxiliary system (18). It is seen that the error z_{i1} evolves within the enlarged constraint spaces from

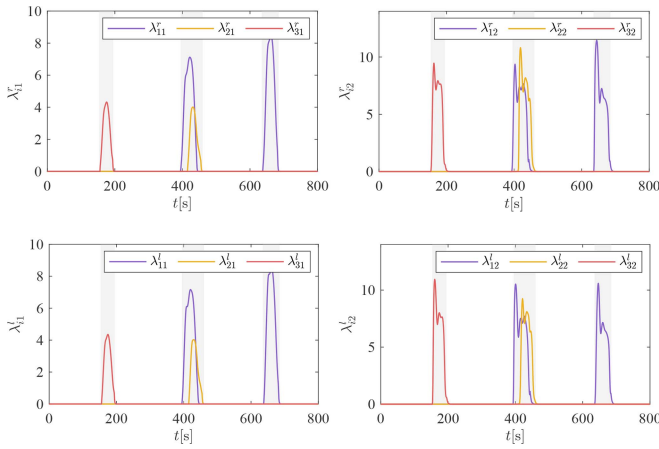


Fig. 7. Auxiliary variables for relative distance and heading errors.

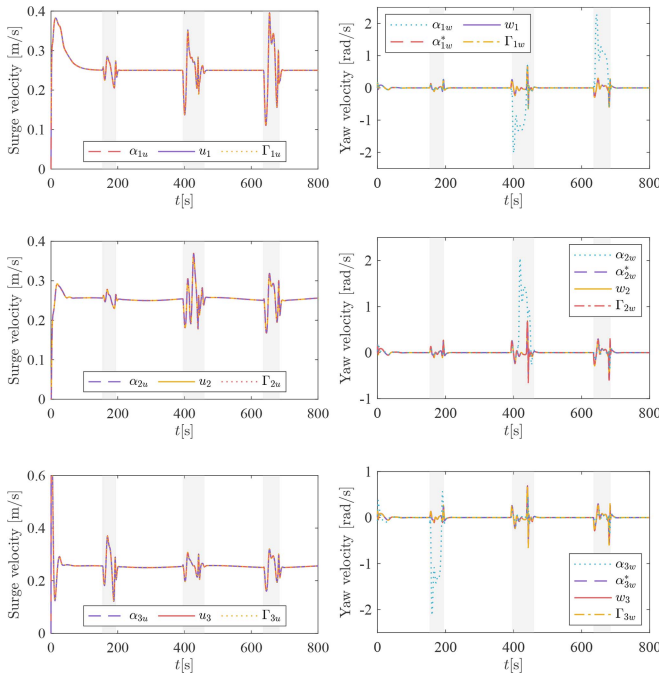


Fig. 8. Surge (left column) and yaw (right column) velocities of all ASVs.

Fig. 6(a), (c), and (e). In addition, it knows that errors z_{12} and z_{22} gently to the prescribed neighboring region of the origin without obvious overshoots from Fig. 6(b), (d), and (f). Fig. 7 draws the modified variables λ_{i1}^r , λ_{i1}^l , λ_{i2}^r , and λ_{i2}^l for relative distances z_{i1} and heading errors z_{i2} , respectively. When it does not avoid collision, the modified variables are at the origin, which means that prescribed performances are not affected. When avoiding collision, the positive modified values are generated to enhance the adaptability of prescribed bounds without changing parameters $\rho_{i1,\infty}$ and $\rho_{i2,\infty}$.

Fig. 8 gives the surge and yaw velocity signals of ASV1–ASV3, respectively. According to Fig. 8(b), the yaw velocity adjustment in the first gray bar is to hold the cooperation of ASV1–ASV3 when ASV3 avoids Obstacle1, and the adjustments of the second and third gray bars are due to avoiding Obstacle2 and intersection ASV for ASV1. In Fig. 9, the disturbance estimations for ASV1 in the surge and yaw

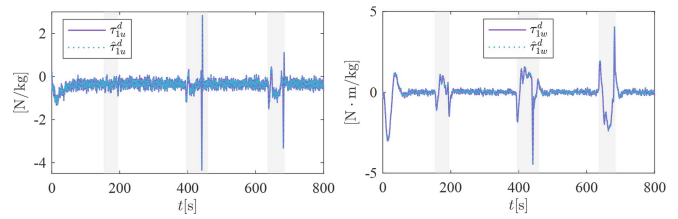


Fig. 9. Estimation performances of PTDO for ASV1.

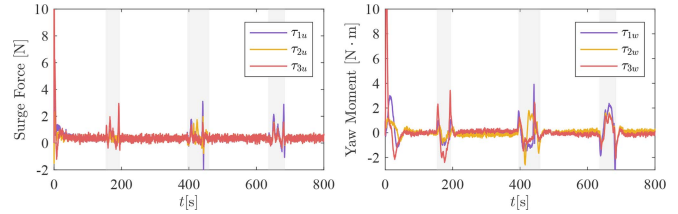


Fig. 10. Properties of control inputs.

direction are presented by using the proposed PTDOs. Fig. 10 shows the surge forces and yaw moments of ASV1–ASV3.

VI. CONCLUSION

In this article, we introduce a constrained safe cooperative maneuvering method for multiple underactuated ASVs subject to performance-prescribed and obstacle-loaded constraints. Our proposed method contains a guidance loop and a control loop. In the guidance loop, the designed ATPP can not only achieve the transient and steady-state indices of relative position and heading control but also modify bounds for possible collision avoidance actions of ASVs. The barrier-certified yaw velocity protocol guarantees the safety of multi-ASV system in an obstacle-loaded environment. In the control loop, the PTDO-based kinetic control laws are devised under unknown environmental disturbances. Simulation results of three ASVs verified the efficacy of the constrained safe cooperative maneuvering method. For future study, it is desirable to achieve the constrained safe cooperative control of ASVs with the Convention on the International Regulations for Preventing Collisions at Sea (COLREGs).

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